

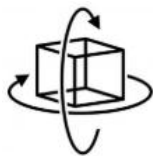
## Bluetooth AHRS IMU sensor | BWT901CL

*The Robust Acceleration, Angular velocity, Angle & Magnetic filed Detector*

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*The BWT901CL is a Bluetooth 2.0 multi-sensor device, detecting acceleration, angular velocity, air pressure angle as well as magnetic filed. The robust housing and the small outline makes it perfectly suitable for industrial applications such as condition monitoring and predictive maintenance. Configuring the device enables the customer to address a broad variety of application by interpreting the sensor data by smart algorithms and Kalman filtering.*

### BUILT-IN SENSORS



Accelerometer



Gyroscope



Magnetometer



## Tutorial Link

[Google Drive](#)

**Link to instructions DEMO:**

[WITMOTION Youtube Channel](#)

[BWT901BCL Playlist](#)

If you have technical problems or cannot find the information that you need in the provided documents, please contact our support team. Our engineering team is committed to providing the required support necessary to ensure that you are successful with the operation of our AHRS sensors.

## Contact

[Technical Support Contact Info](#)

## Application

- AGV Truck
- Platform Stability
- Auto Safety System
- 3D Virtual Reality
- Industrial Control
- Robot
- Car Navigation
- UAV
- Truck-mounted Satellite Antenna Equipment

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# 1 Overview

BWT901CL's scientific name is AHRS IMU sensor. A sensor measures 3-axis angle, angular velocity, acceleration, magnetic field. Its strength lies in the algorithm which can calculate three-axis angle accurately.

BWT901BCL is an CE certified accelerometer. It is employed where the highest measurement accuracy is required. BWT901BCL offers several advantages over competing sensor:

- Heated for best data availability: new WITMOTION patented zero-bias automatic detection calibration algorithm outperforms traditional accelerometer sensor
- High precision Roll Pitch Yaw (X Y Z axis) Acceleration + Angular Velocity + Angle + Air Pressure + Magnetic Field output
- Low cost of ownership: remote diagnostics and lifetime technical support by WITMOTION service team
- Developed tutorial: providing manual, datasheet, Demo video, free software for Windows computer, APP for Android smartphones , and sample code for MCU integration including 51 serial, STM32, Arduino, Matlab, Raspberry Pi, communication protocol for project development
- WITMOTION sensors have been praised by thousands of engineers as a recommended attitude measurement solution



## 2 Features

- Built-in BWT901B module, for detailed parameters, please refer to the instructions.
- The baud rate of this device is 115200 and cannot be changed.
- The interface of this product only leads to a serial port and Bluetooth EDR,
- The module consists of a high precision gyroscope, accelerometer and geomagnetic field sensor. The product can solve the current real-time motion posture of the module quickly by using the high-performance microprocessor, advanced dynamic solutions and Kalman filter algorithm.
- The advanced digital filtering technology of this product can effectively reduce the measurement noise and improve the measurement accuracy.
- Maximum 200Hz data output rate. Output content can be arbitrarily selected, the output speed 0.2HZ~ 200HZ adjustable.

## 3 Specification

### 3.1 Parameter

Parameter	Specification
➤ Voltage	3.3V-5V
➤ Current	<40mA
➤ Battery	250mAh, 3.7V
➤ Working hour	A. Play 4h at 1 charge (battery) B. Power source of 5V
➤ Size	51.3mm x36mm X15mm/ 2.02" x 1.41" x 0.59"
➤ Data	Angle: X Y Z, 3-axis Acceleration: X Y Z, 3-axis Angular Velocity: X Y Z, 3-axis Magnetic Field : X Y Z, 3-axis Time, Quaternion Air Pressure
➤ Output frequency	0.2Hz--200Hz
➤ Interface	Serial TTL level, Baud rate--115200 (default, can not be changed ) Interface standard: Type-C
➤ Bluetooth	Range: ≤10m Bluetooth module: HC-06 Version: EDR Bluetooth 2.0

### Measurement Range & Accuracy

Sensor	Measurement Range	Accuracy/ Remark
➤ Accelerometer	X, Y, Z, 3-axis: ±16g	Accuracy: 0.01g Resolution: 16bit Stability: 0.005g
➤ Gyroscope	X, Y, Z, 3-axis ±2000°/s	Resolution: 16bit Stability: 0.05°/s
➤ Magnetometer	X, Y, Z, 3-axis ±4900μT	0.15μT/LSB typ. (16-bit)
➤ Angle/ Inclinometer	X, Y, Z, 3-axis X, Z-axis: ±180° Y ±90° (Y-axis 90° is singular point)	Accuracy: X, Y-axis: 0.05° Z-axis: 1°(after magnetic calibration)
➤ Barometer	1 axis	



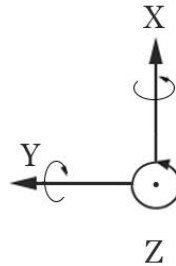
### 3.2 Size



Parameter	Specification	Tolerance	Comment
Length	51.3	$\pm 0.2$	Unit: millimeter.
Width	36	$\pm 0.2$	
Height	15	$\pm 0.2$	
Weight	20	$\pm 0.2$	Unit: gram

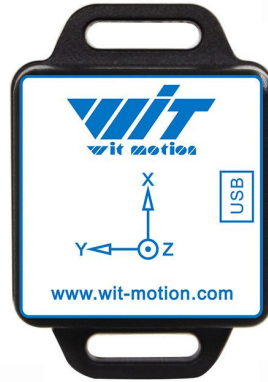
### 3.3 Axial Direction

The coordinate system used for attitude angle settlement is the northeast sky coordinate system. Place the module in the positive direction, as shown in the figure below, direction forward is the X-axis, the direction left is the Y-axis, and direction upward is the Z-axis. Euler angle represents the rotation order of the coordinate system when the attitude is defined as Z-Y-X, that is, first turn around the Z-axis, then turn around the Y-axis, and then turn around the X-axis. Counterclockwise rotation is positive.



## 4 Pin Definition

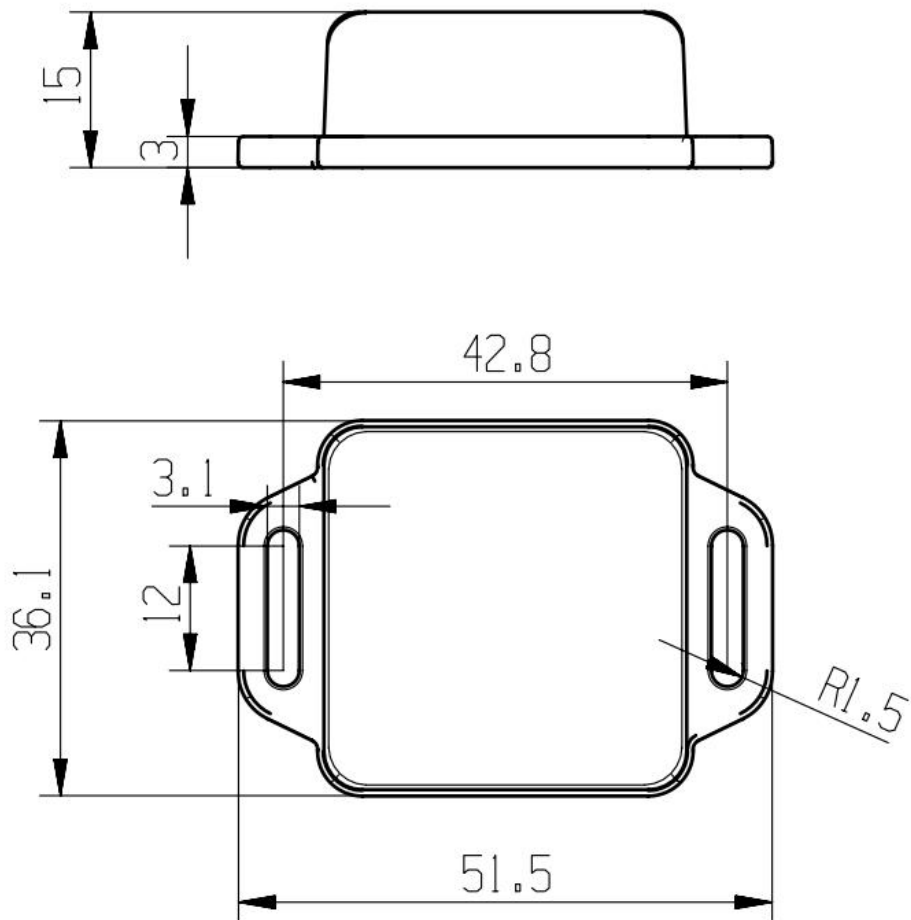
### Interface Standard



Type-C

PIN	Function
➤ Type-C	3.3-5V input supply Wired Connection with TTL Singal Communication

## 5 Casing Specification





## 6 Communication Protocol

Level: TTL level

Baud rate: 115200 (default, cannot be changed), stop bit and parity bit 0

### 6.1 Output Data Format

#### 6.1.1 Time Output

0x55	0x50	YY	MM	DD	hh	mm	ss	msL	msH	SUM
------	------	----	----	----	----	----	----	-----	-----	-----

YY: Year, 20YY Year

MM: Month

DD: Day

hh: hour

mm: minute

ss: Second

ms: Millisecond

Millisecond calculate formula:

$ms = ((msH < 8) | msL)$

$Sum = 0x55 + 0x51 + YY + MM + DD + hh + mm + ss + ms + VL$



## 6.1.2 Acceleration Output

0x55	0x51	AxL	AxH	AyL	AyH	AzL	AzH	TL	TH	SUM
------	------	-----	-----	-----	-----	-----	-----	----	----	-----

Calculate formula:

$$a_x = ((AxH < 8) | AxL) / 32768 * 16g \text{ (g is Gravity acceleration, } 9.8m/s^2)$$

$$a_y = ((AyH < 8) | AyL) / 32768 * 16g \text{ (g is Gravity acceleration, } 9.8m/s^2)$$

$$a_z = ((AzH < 8) | AzL) / 32768 * 16g \text{ (g is Gravity acceleration, } 9.8m/s^2)$$

Temperature calculated formula:

$$T = ((TH < 8) | TL) / 100 \text{ } ^\circ C$$

Checksum:

$$Sum = 0x55 + 0x51 + AxH + AxL + AyH + AyL + AzH + AzL + TH + TL$$

Note:

1. The data is sent in hexadecimal, not ASCII code.

Each data is transmitted in turn of low byte and high byte, and the two are combined into a signed short type data.

For example, X-axis acceleration data Ax, where AxL is low byte and AxH is high byte. The conversion method is as follows:

Assuming that Data is actual data, DataH is its high byte, and DataL is its low byte, then: Data = (short) (DataH < 8 | DataL).

It must be noted that DataH needs to be coerced into a signed short data and then shifted, and the data type of Data is also a signed short type, so that it can represent negative numbers.

## 6.1.3 Angular Velocity Output

0x55	0x52	wxL	wxH	wyL	wyH	wzL	wzH	VL	VH	SUM
------	------	-----	-----	-----	-----	-----	-----	----	----	-----

Calculated formula:

$$w_x = ((wxH < 8) | wxL) / 32768 * 2000 \text{ (} ^\circ/s)$$

$$w_y = ((wyH < 8) | wyL) / 32768 * 2000 \text{ (} ^\circ/s)$$

$$w_z = ((wzH < 8) | wzL) / 32768 * 2000 \text{ (} ^\circ/s)$$

Battery Voltage calculated formula:

$$\text{Battery Voltage} = ((VH < 8) | VL) / 100 \text{ V}$$

Checksum:

$$Sum = 0x55 + 0x52 + wxH + wxL + wyH + wyL + wzH + wzL + VH + VL$$



## 6.1.4 Angle Output

0x55	0x53	RollL	RollH	PitchL	PitchH	YawL	YawH	VL	VH	SUM
------	------	-------	-------	--------	--------	------	------	----	----	-----

Calculated formula:

Roll(X axis) $Roll = ((RollH << 8) | RollL) / 32768 * 180(^{\circ})$

Pitch(Y axis) $Pitch = ((PitchH << 8) | PitchL) / 32768 * 180(^{\circ})$

Yaw(Z axis) $Yaw = ((YawH << 8) | YawL) / 32768 * 180(^{\circ})$

Version calculated formula:

Version $Version = (VH << 8) | VL$

Checksum:

Sum $Sum = 0x55 + 0x53 + RollH + RollL + PitchH + PitchL + YawH + YawL + VH + VL$

Note:

1. The coordinate system used for attitude angle settlement is the northeast sky coordinate system. Place the module in the positive direction, as the figure shown in Chapter 3.3, direction left is the Y-axis, the direction forward is the X-axis, and direction upward is the Z-axis. Euler angle represents the rotation order of the coordinate system when the attitude is defined as Z-Y-X, that is, first turn around the Z-axis, then turn around the Y-axis, and then turn around the X-axis.
2. Although the range of the roll angle is  $\pm 180$  degrees, in fact, since the coordinate rotation sequence is Z-Y-X, when expressing the attitude, the range of the pitch angle (Y-axis) is only  $\pm 90$  degrees, and it will change to less than 90 after exceeding 90 degrees Degrees while making the X-axis angle greater than 180 degrees. For detailed principles, please Google Euler angle and posture-related information.
3. Since the three axes are coupled, they will show independent changes only at small angles, and the attitude angles will change at large angles. For example, when the Y-axis is close to 90 degrees, even if the attitude only rotates around the Y-axis, the angle of the axis will also change greatly, which is an inherent problem with Euler angles indicating attitude.

## 6.1.5 Magnetic Output

0x55	0x54	HxL	HxH	HyL	HyH	HzL	HzH	TL	TH	SUM
------	------	-----	-----	-----	-----	-----	-----	----	----	-----

Calculated formula:

Magnetic(x axis) $H_x = ((H_{xH} \ll 8) | H_{xL})$

Magnetic(y axis) $H_y = ((H_{yH} \ll 8) | H_{yL})$

Magnetic(z axis) $H_z = ((H_{zH} \ll 8) | H_{zL})$

Temperature calculated formula:

$T = ((T_H \ll 8) | T_L) / 100^\circ\text{C}$

Checksum:

$\text{Sum} = 0x55 + 0x53 + H_{xH} + H_{xL} + H_{yH} + H_{yL} + H_{zH} + H_{zL} + T_H + T_L$

## 6.1.6 Quaternion

0x55	0x59	Q0L	Q0H	Q1L	Q1H	Q2L	Q2H	Q3L	Q3H	SUM
------	------	-----	-----	-----	-----	-----	-----	-----	-----	-----

Calculated formula:

$Q_0 = ((Q_{0H} \ll 8) | Q_{0L}) / 32768$

$Q_1 = ((Q_{1H} \ll 8) | Q_{1L}) / 32768$

$Q_2 = ((Q_{2H} \ll 8) | Q_{2L}) / 32768$

$Q_3 = ((Q_{3H} \ll 8) | Q_{3L}) / 32768$

Checksum:

$\text{Sum} = 0x55 + 0x59 + Q_{0L} + Q_{0H} + Q_{1L} + Q_{1H} + Q_{2L} + Q_{2H} + Q_{3L} + Q_{3H}$



## 6.2 Config Commands

Reminder:

1. Data format

0xFF	0xAA	Address	DataL	DataH
------	------	---------	-------	-------

### 6.2.1 Register Address

Address	Symbol	Meaning
0x00	SAVE	Save
0x01	CALSW	Calibration
0x02	RSW	Return data content
0x03	RATE	Return data Speed
0x04	BAUD	Baud rate
0x05	AXOFFSET	X axis Acceleration bias
0x06	AYOFFSET	Y axis Acceleration bias
0x07	AZOFFSET	Z axis Acceleration bias
0x08	GXOFFSET	X axis angular velocity bias
0x09	GYOFFSET	Y axis angular velocity bias
0x0a	GZOFFSET	Z axis angular velocity bias
0x0b	HXOFFSET	X axis Magnetic bias
0x0c	HYOFFSET	Y axis Magnetic bias
0x0d	HZOFFSET	Z axis Magnetic bias
0x1a	IICADDR	IIC address
0x1b	LEDOFF	Turn off LED
0x1c	GPSBAUD	GPS baud rate
0x30	MMYY	Month , Year
0x31	HHDD	Hour , Day
0x32	SSMM	Second , Minute
0x33	MS	Millisecond
0x34	AX	X axis Acceleration
0x35	AY	Y axis Acceleration
0x36	AZ	Z axis Acceleration
0x37	GX	X axis angular velocity
0x38	GY	Y axis angular velocity
0x39	GZ	Z axis angular velocity
0x3a	HX	X axis Magnetic
0x3b	HY	Y axis Magnetic



0x3c	HZ	Z axis Magnetic
0x3d	Roll	X axis Angle
0x3e	Pitch	Y axis Angle
0x3f	Yaw	Z axis Angle
0x40	TEMP	Temperature
0x51	Q0	Quaternion Q0
0x52	Q1	Quaternion Q1
0x53	Q2	Quaternion Q2
0x54	Q3	Quaternion Q3

## 6.2.2 Save Configuration

0xFF	0xAA	0x00	SAVE	0x00
------	------	------	------	------

SAVE: Save

- 0: Save current configuration
- 1: set to default setting

## 6.2.3 Calibrate

0xFF	0xAA	0x01	CALSW	0x00
------	------	------	-------	------

CALSW: Set calibration mode

- 0: Exit calibration mode
- 1: Enter Gyroscope and Accelerometer calibration mode
- 2: Enter magnetic calibration mode

## 6.2.4 Installation Direction

0xFF	0xAA	0x23	DIRECTION	0x00
------	------	------	-----------	------

DIRECTION: set installation direction

- 0: set to horizontal installation
- 1: set to vertical installation

## 6.2.5 Sleep/ Wake up

0xFF	0xAA	0x22	0x01	0x00
------	------	------	------	------

After sending the command, the module enters the sleep (standby) state, and once again, the module enters the working state from the standby state.

## 6.2.6 Algorithm Transition

0xFF	0xAA	0x24	ALG	0x00
------	------	------	-----	------

ALG: 6-axis/ 9-axis algorithm transition

0: switch to 9-axis algorithm

1: switch to 6-axis algorithm

## 6.2.7 Gyroscope Automatic Calibration

0xFF	0xAA	0x63	GYRO	0x00
------	------	------	------	------

GYRO: gyroscope automatic calibration

0: set to gyroscope automatic calibration

1: removed to gyroscope automatic calibration

## 6.2.8 Return Content

0xFF	0xAA	0x02	RSWL	RSWH
------	------	------	------	------

RSWL byte definition

byte	7	6	5	4	3	2	1	0
Name	0x57 pack	0x56 pack	0x55 pack	0x54 pack	0x53 pack	0x52 pack	0x51 pack	0x50 pack
default	0	0	0	1	1	1	1	0

RSWH byte definition

byte	7	6	5	4	3	2	1	0
Name	X	X	X	X	X	0x5A pack	0x59 pack	0x58 pack
default	0	0	0	0	0	0	0	0

X is an undefined value.

0x50 pack: time pack

0: Not output 0X50 pack

1: Output 0X50 pack

0x51 pack: Acceleration pack

0: Not output 0x51 pack

1: Output 0x51 pack

0x52 pack: Angular velocity pack

0: Not output 0x52 packet

1: Output 0x52 pack

0x53 pack: Angle Pack

0: Not output 0x53 pack

1: Output 0x53 pack

0x54 pack: Magnetic Pack

0: Not output 0x54 pack

1: Output 0x54 pack

0x59 pack: Quaternion Pack

0: Not output 0x59 pack

1: Output 0x59 pack



## 6.2.9 Return Rate

0xFF	0xAA	0x03	RATE	0x00
------	------	------	------	------

RATE: return rate

0x01 :0.2Hz

0x02: 0.5Hz

0x03: 1Hz

0x04: 2Hz

0x05: 5Hz

0x06: 10Hz(default)

0x07: 20Hz

0x08: 50Hz

0x09: 100Hz

0x0a: 125Hz

0x0b: 200Hz

0x0c: Single

0x0d: Not output

After the setup is complete, need to click save, and re-power the module to take effect.

## 6.2.10 Set X Axis Acceleration Bias

0xFF	0xAA	0x05	AXOFFSETL	AXOFFSETH
------	------	------	-----------	-----------

AXOFFSETL: X axis Acceleration bias low byte

AXOFFSETH: X axis Acceleration bias high byte

AXOFFSET= (AXOFFSETH <<8) | AXOFFSETL

Note: After setting the acceleration bias, the output value of the acceleration is the sensor measured value minus the bias value.

## 6.2.11 Set Y Axis Acceleration Bias

0xFF	0xAA	0x06	AYOFFSETL	AYOFFSETH
------	------	------	-----------	-----------

AYOFFSETL: Y axis Acceleration bias low byte

AYOFFSETH: Y axis Acceleration bias high byte

AYOFFSET= (AYOFFSETH <<8) | AYOFFSETL

Note: After setting the acceleration bias, the output value of the acceleration is the sensor measured value minus the bias value.

## 6.2.12 Set Z Axis Acceleration Bias

0xFF	0xAA	0x07	AZOFFSETL	AZOFFSETH
------	------	------	-----------	-----------

AZOFFSETL: Z axis Acceleration bias low byte

AZOFFSETH: Z axis Acceleration bias high byte

AZOFFSET= (AZOFFSETH <<8) | AZOFFSETL

Note: After setting the acceleration bias, the output value of the acceleration is the sensor measured value minus the bias value.

### 6.2.13 Set X Axis Angular Velocity Bias

0xFF	0xAA	0x08	GXOFFSETL	GXOFFSETH
------	------	------	-----------	-----------

GXOFFSETL: Set X axis Angular velocity bias low byte

GXOFFSETH: Set Y axis Angular velocity bias high byte

$GXOFFSET = (GXOFFSETH \ll 8) | GXOFFSETL$

Note: After setting the angular velocity zero deviation, the output value of the angular velocity is the sensor measurement value minus the zero deviation value.

### 6.2.14 Set Y Axis Angular Velocity Bias

0xFF	0xAA	0x09	GYOFFSETL	GYOFFSETH
------	------	------	-----------	-----------

GYOFFSETL: Set X axis Angular velocity bias low byte

GYOFFSETH: Set X axis Angular velocity bias high byte

$GYOFFSET = (GYOFFSETH \ll 8) | GYOFFSETL$

Note: After setting the angular velocity zero deviation, the output value of the angular velocity is the sensor measurement value minus the zero deviation value.

### 6.2.15 Set Z Axis Angular Velocity Bias

0xFF	0xAA	0x0a	GZOFFSETL	GZOFFSETH
------	------	------	-----------	-----------

GZOFFSETL: Set Z axis Angular velocity bias low byte

GZOFFSETH: Set Z axis Angular velocity bias low byte

$GZOFFSET = (GZOFFSETH \ll 8) | GZOFFSETL$

Note: After setting the angular velocity zero deviation, the output value of the angular velocity is the sensor measurement value minus the zero deviation value.



## 6.2.16 Set X Axis Magnetic Bias

0xFF	0xAA	0x0b	HXOFFSETL	HXOFFSETH
------	------	------	-----------	-----------

HXOFFSETL: Set X axis magnetic bias low byte

HXOFFSETH: Set X axis magnetic bias high byte

$HXOFFSET = (HXOFFSETH \ll 8) | HXOFFSETL$

Note: After setting the magnetic field bias, the output value of the magnetic field is the sensor measured value minus the zero bias value.

## 6.2.17 Set Y Axis Magnetic Bias

0xFF	0xAA	0x0c	HXOFFSETL	HXOFFSETH
------	------	------	-----------	-----------

HXOFFSETL: Set Y axis magnetic bias low byte

HXOFFSETH: Set Y axis magnetic bias high byte

$HXOFFSET = (HXOFFSETH \ll 8) | HXOFFSETL$

Note: After setting the magnetic field bias, the output value of the magnetic field is the sensor measured value minus the zero bias value.

## 6.2.18 Set Z Axis Magnetic Bias

0xFF	0xAA	0x0d	HXOFFSETL	HXOFFSETH
------	------	------	-----------	-----------

HXOFFSETL: Set Y axis magnetic bias low byte

HXOFFSETH: Set Z axis magnetic bias high byte

$HXOFFSET = (HXOFFSETH \ll 8) | HXOFFSETL$

Note: After setting the magnetic field bias, the output value of the magnetic field is the sensor measured value minus the zero bias value.

## 6.2.19 LED

0xFF	0xAA	0x1b	LEDOFF	0x00
------	------	------	--------	------

LEDOFF:

0x01: Turn off LED

0x00: Turn on LED

## 7. Certification

Shenzhen Tongzhou Testing Co.,Ltd.

### EC CERTIFICATE

**Certificate No.** : TZ200301235-C

**Applicant** : WitMotion Shenzhen Co.,Ltd

**Address** : West Industrial Park Tantou 10th building B Building 2nd Floor Songgang Town Baoan District Shenzhen City Guangdong Province

**Product** : 9-axis Bluetooth 2.0 Attitude Sensor

**Trademark** : WITMOTION

**Model(s)** : BWT901CL, BWT901BCL, BWT61PCL, BWT61CL, BWT61, BWT901, BWT61, BWT901

**Manufacturer** : WitMotion Shenzhen Co.,Ltd

**Address** : West Industrial Park Tantou 10th building B Building 2nd Floor Songgang Town Baoan District Shenzhen City Guangdong Province

Essential Requirement	Applied Specifications/ Standards	Documentary Evidence	Result
EMC	Draft ETSI EN 301 489-1 V2.2.0 (2017-03) Draft ETSI EN 301 489-17 V3.2.0 (2017-03) EN 55032: 2015 EN 55035: 2017	Report TZ200301235-RE	Conform
Radio	ETSI EN 300 328 V2.1.1 (2016-11)	Report TZ200301235-EDR	Conform
Health	EN 62479:2010	Report TZ200301235-EMF	Conform
Safety	EN 62368-1: 2014	Report TZ200301235-S	Conform

The EUT described above has been tested by us with the listed standards and found in compliance with the council RED 2014/53/EU. It is possible to use CE marking to demonstrate the compliance with the Directive. The scope of evaluation relates to the submitted documents only.





*Andy Zhang*  
For Chief Executive  
April 3, 2020

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